

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Repaired damage to aircraft by replacing the following components, using Beech Aircraft factory replacement parts:

1 ea.	3A32C76/82NB-2 Propeller.		
1 "	35-110061-1 Spar Assy., front L.H. Wing		
1 "	35-110009-603 Leading Edge, L.H. wing outbd.		
1 "	35-170005-19 Wing Tip, L.H.		
1 "	35-415343-608 Keel, L.H.	1 ea.	35-115130-602 Rib
1 "	35-415343-609 " , R.H.	1 "	35-115032-602 "
1 "	35-410444-54 Keel Assy.	1 "	96-110007-610 "
1 "	35-115137-22 Skin	1 "	35-115381-1 Fillet
1 "	35-410452-79 "	1 "	35-364259-1 Bracket
1 "	35-410452-190 "	1 "	95-824018 Brace
1 "	35-410452-191 "	1 "	95-824005-601 "
1 "	35-410452-7 "	1 "	35-410077-30 Door
1 "	35-410452-115 "	1 "	35-410077-31 "
1 "	35-115199-24 "	1 "	35-910321-608 "
1 "	35-410000-625 "	1 "	35-815055-69 "
1 "	50-110010-128 "	1 "	35-950005-9 Tube
1 "	35-415373-602 Bulkhead Assy.	1 "	35-550567-27 Duct
1 "	35-410215-44 "	1 "	35-550567-29 "
1 "	35-410215-5 "	1 "	35-910053-1 Flap Assy.
1 "	35-415373-602 "	1 "	35-115382-1 Tip Assy.
1 "	35-400364-10 Channel	1 "	HM4U Rod End
1 "	95-420015-15 "	1 "	35-820107-05 Knee
1 "	35-400366-3 Support	1 "	35-825193-1 "
1 "	35-410210-4 Frame	1 "	96-364012-5 Cla mp
2 "	35-410000-123 Stiffener	8 "	35-115388-4 Pin, hinge
1 "	35-410000-47 Stringer	1 "	35-825087-16 Rod
1 "	35-410000-36 "	1 "	35-410215-5 Blkhd., L.H.
1 "	35-410000-25 "	1 "	35-400364-610 Channel, R.H.
1 "	35-410444-24 Angle	1 "	35-115031-1 Rib Assy., R.H.
1 "	35-410035-1 "	1 "	95-420015-45 Stringer, R.H.
2 "	35-410279-1 "	1 "	95-420015-139 Frame
1 "	35-410461-5 "	1 "	35-115431-109 Hinge, upper inbd.

A new cabin bottom aft center skin, Beech P/N 35-410452-183, was installed from fuselage station 85 to station 118, and was fabricated of .020" 2024T3 alclad aluminum sheet. The left wing bottom outboard skin, Beech P/N 35-115199-27, and bottom center skin, Beech P/N 35-115200-107, were replaced and were fabricated of .020" 2024T3 alclad aluminum sheet. All fabricated skins were same thickness as original except for left bottom center wing skin which was increased from .016" to .020" thickness.

All work was accomplished in accordance with AC43.13-1, chapter 2, sections 1 and 3, chapter 5 and chapter 11. No weight change was involved.

- END -

ADDITIONAL SHEETS ARE ATTACHED